GNSS Array-Based Ionospheric Spatial Gradient Monitoring: Precision and Integrity Analysis

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Biography

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Abstract

Integrity monitoring and reliable detection of the ionospheric spatial gradient is of importance for various fields including total electron content modeling, radio communication and radar, and differential GNSS applications such as landing, departure and surface operations at an airport. In this contribution, we investigate the precision and the integrity of the ionospheric spatial gradient estimation upon the double-differenced (DD) array-based GNSS model of observations. Closed-form expressions for the variance matrix of the ionospheric spatial gradient and the corresponding uniformly most powerful invariant (UMPI) minimal detectable biases (MDBs) are provided for different scenarios, through which we assess the importance of several contributing factors. The closed-form expressions will show that the precision of the ionospheric spatial gradient estimator and its detectability are independent of our choice for the reference receiver and the reference satellite when forming the DD equations.

In this contribution it will be shown how the precision of the estimated ionospheric spatial gradient and its MDB depend on and benefit from the number of receivers, number of satellites, number of frequencies, and the size and geometry of the array.

1 Introduction

Acting as a dispersive environment, the ionosphere affects the GNSS measurements through introducing a delay in the code signals and an advance in the carrier-phase signals, see e.g. [1, 2]. Since the ionosphere varies in both space and time, the affected GNSS signals would therefore also experience spatial and temporal variations. Large values for the ionospheric spatial gradient are expected when an ionospheric front occurs. The possible undesirable impact of the ionospheric spatial gradients needs therefore to be properly taken into account. Indeed, integrity monitoring and reliable detection of the ionospheric spatial gradient have recently been given growing attention in various disciplines such as estimating and modeling total electron content [3–5], radio communication and radar [6], and differential GNSS applications such as landing, departure and surface operations at an airport [7–10]. In case of the aviation applications for instance, a single Ground Facility (GF), comprising multiple spatially-separated GNSS receivers, is meant to ensure safe flight operations through monitoring signal-in-space failures, e.g. ionospheric spatial gradient that are potentially hazardous for the user (aircraft) positioning.

In this contribution, we study the precision and integrity of the ionospheric spatial gradient estimation using an array of GNSS antennas. Our study characterizes the role of contributing factors including number of antennas, number of frequencies, range of frequencies, configuration of the GF antennas, number of satellites, and satellites elevation. Of many detailed studies regarding the ionospheric spatial gradient monitoring techniques, there are quite recent contributions that are also dedicated to the array-based estimation/detection of the ionospheric spatial gradient, see e.g. [7, 11–14]. Our study differs from those contributions in the following four aspects:

1) While multiple baselines are utilized, our approach to the precision and reliability analysis of the ionospheric
spatial gradient remains invariant for the choice of reference antenna, thereby being independent of how receiver pairs are selected to form the baselines, see e.g. [12, 14].  

2) Although GBAS is currently operational on a single frequency L1, it is going to be modernized to multiple frequencies [15, 16]. Therefore, our analysis is not restricted to the single-frequency GBAS scenario, providing an opportunity to characterize the impact of the number/structure of frequencies on the detection of the ionospheric spatial gradient.

3) In the previous studies, attention is focused on the estimation/detection of the ionospheric spatial gradient based on the observations of a single satellite pair [12–14]. In this study, we relax this assumption by considering the contribution of multiple satellites to the detection of ionospheric spatial gradient. The gradient detection of a single satellite pair would then follow as a special case.

4) We assess how including the observations of more than one epoch contributes to the detection procedure of the ionospheric spatial gradient, for both ambiguity-float case and ambiguity-fixed case. The impact of the aforementioned contributing factors on integer ambiguity resolution success rate will be investigated as well.

In the present contribution, the precision of the array-based ionospheric spatial gradient estimator will be studied through the analytical structure of its variance matrix, whereas its integrity will be studied by means of its minimal detectable bias (MDB) [17, 18]. The MDBs in this contribution are associated with the uniformly most powerful invariant (UMPI) test [19, 20]. Closed-form expressions of the stated variance matrix and MDBs will be presented as functions of the contributing factors, e.g. number of antennas and number of frequencies. This gives further insights into the role of these contributing factors and allows one to conduct an a-priory analysis before an actual measurement campaign is launched. Our analytical formulation will be accompanied by numerical results.

2 Ionospheric spatial gradient monitoring

2.1 Array-based GNSS model of observations

We assume that the GNSS signals transmitted by m satellites on f frequencies, are tracked by n receivers. If the relative receivers positions and tropospheric delays are known, the multivariate double-differenced (DD) GNSS observation equations can be written in the compact form,

$$ E \left[ \begin{array}{l} \phi_k^s \\ \phi_k^q \\ \end{array} \right] = \left[ \begin{array}{cc} I_{n-1} \otimes \mu \otimes I_{n-1} & 0 \\ -I_{n-1} \otimes \mu \otimes I_{n-1} & I_{n-1} \otimes \Lambda \otimes I_{n-1} \end{array} \right] \left[ \begin{array}{l} \phi_k^s \\ \phi_k^q \end{array} \right] $$

where $E[.]$ is the expectation operator, and $\otimes$ denotes the Kronecker product [21]. Considering $p$ and $q$ as, respectively, the pivot receiver and the pivot satellite, the DD ‘observed-minus-computed’ code observations associated with $n$ receivers, satellite $s$ and satellite $q$ on frequency $f$ are stacked in the $(n-1)$-vector $p_k^{sf} = [p_{p1,f}^{sf}, \ldots, p_{pn,f}^{sf}]^T$.

The $f(n-1)$-vector $p_k^{sf} = [p_{p1,f}^{sf}, \ldots, p_{pm,f}^{sf}]^T$ then contains all the DD code observations associated with satellites $s$ and $q$. The $f(n-1)(m-1)$-vector of the DD code observations is structured as $p_k^{sf} = [p_{p1}^{sfT}, \ldots, p_{pm}^{sfT}]^T$. The $f(n-1)(m-1)$-vector of phase observations is obtained likewise. The $f(n-1)(m-1)$-vector of integer-valued DD ambiguities $s_k^{sf}$, in cycles, has the same order as $p_k^{sf}/\lambda_f T$. $s_k^{sf}$ is the $(n-1)(m-1)$-vector of the DD first-order ionospheric delays on the first frequency. $I_f$ is the identity matrix of size $f$. $\Lambda = \text{diag}(\lambda_1, \ldots, \lambda_f)$ with $\lambda_i$ the wavelength of frequency $f_i$. The $f$-vector $\mu$ includes the frequency-dependent ratios $\mu_i = \lambda_f^2/\lambda_i^2$.

The stochastic model is assumed given as

$$ D \left[ \begin{array}{l} \phi_k^s \\ \phi_k^q \end{array} \right] = \left[ \begin{array}{ll} \sigma_s & 0 \\ 0 & \sigma_q \end{array} \right] \otimes D_n^{w^{-1}} D_m \otimes I_f \otimes D_f^s $$

where $D[.]$ is the dispersion operator. The $(m-1) \times m$ matrix $D_f^s$ and $(n-1) \times n$ matrix $D_n^{w^{-1}}$ are the differencing operators. $\sigma_s$ and $\sigma_q$ are, respectively, the zenith-referenced standard deviation of the undifferenced code and phase observables. $W = \text{diag}(w_1, \ldots, w_m)$ contains the satellite-elevation-dependent weights $w_i$. In this paper, we use the following form [22]

$$ w_i = [1 + 10 \exp(-(e_i^s/10))]^{-2} $$

with $e_i^s$ being the elevation of satellite $i$ in degrees. If $e_i^s = 90^\circ$, then $w_i$ is in agreement with 1 up to 2 digits after the decimal point.

Our aim, in this study, is to detect the ionospheric spatial gradient using an array of GNSS antennas. When we speak of an array of antennas, it means that the antennas are located within few kilometers of each other. For such a small-size array, one can make some assumptions leading to a more simplified model of observations to work with. One of the possible suppositions is that signals from a particular GNSS satellite to all the antennas existing in the array experience almost the same ionospheric delays under nominal ionosphere condition (no ionospheric front or storm), leading to the ionospheric delays becoming independent of the receiver location. Under these circumstances, the DD ionospheric delays are zero. Hence, with $s_k^{sf} = 0$, our working hypothesis $H_0$ follows then from Eq. (1) as

$$ E \left[ \begin{array}{l} \phi_k^s \\ \phi_k^q \end{array} \right]_{H_0} = \left[ \begin{array}{l} 0 \\ 0 \end{array} \right] $$

where $E[.]$ is the expectation operator, and $\otimes$ denotes the Kronecker product [21].
In the single-epoch case, the number of observations, number of unknowns and the redundancy of the model of observations under $H_a$ are, respectively, $2f(m-1)(n-1)$, $f(m-1)(n-1)$ and $f(m-1)(n-1)$.

In case an ionospheric front affects the signals of one or more satellites, the assumption that the ionospheric delays do not depend on the location of the receivers will not be valid anymore for the contaminated satellites. In previous studies, only the GPS system has been considered, and it is asserted that the probability of a large gradient affecting more than one satellite is sufficiently low, see e.g. [14, 23]. However, when other GNSS systems, e.g. GLONASS and Galileo, are involved, the likelihood of having more than one satellite influenced by the ionospheric front, would increase. Here, in line with previous studies, we consider the case that only one satellite, say $c$, is contaminated by the ionospheric front.

Many studies have considered that the impact of ionospheric delays has both spatial and temporal variations. As mentioned before, ionospheric spatial gradient. As mentioned, ionospheric delays experienced by the signals from satellite $c$ are related to each other through the ionospheric spatial gradient $g$ and the baseline vector between them $b_{12}$.

In Eq. (5), the temporal variation has not been modeled. Figure 1 shows the impact of the ionospheric front on the signals of one of the GNSS satellites as function of the between-antennas baseline vector.

In order to see how the ionospheric spatial gradient affects the DD GNSS observables, we start with the undifferenced level. The ionospheric delays on the first frequency associated with $n$ receivers and satellite $s$ are stacked in the $n$-vector $\tilde{v}_0 = [\tilde{v}_1, \ldots, \tilde{v}_n]^T$. The $nm$-vector of the undifferenced ionospheric delays is structured as $\tilde{v}_R = [\tilde{v}_R^T, \ldots, \tilde{v}_R^T, \ldots, \tilde{v}_R^T]^T$. Applying the DD operator $D_m^T \otimes D_n^T$ on to the both sides of the last equation, those ionospheric delays associated with the uncontaminated satellites will disappear. Therefore we have

$$\tilde{v}_k = (D_m^T u_m^T \otimes D_n^T ) \tilde{v}_k$$  \hspace{1cm} (6)

in which the $m$-vector $u_m^T$ is a canonical unit vector with 1 as its $c^{th}$ element and zeros elsewhere. $\tilde{v}_k$ can be parametrized using Eq. (5) as

$$\tilde{v}_k = e_a \tilde{v}_0 + B g$$  \hspace{1cm} (7)

where the $n \times 3$ matrix $B$ includes the vectors $b_{12}^T$ ($r = 1, \ldots, n$) as its rows, and $e_a$ is the $n$-vector of ones. Combining Eqs. (6) and (7), we get

$$\tilde{v}_k = (D_m^T u_m^T \otimes D_n^T B) g$$  \hspace{1cm} (8)

Since $D_m^T e_n = 0$, $\tilde{v}_0$ is eliminated in Eq. (8). Substituting Eqs. (8) into (1), we obtain the alternative model $H_a$ as

$$E \begin{bmatrix} \tilde{v}_k^T \\ \tilde{v}_0^T \end{bmatrix} |_{H_a} = \begin{bmatrix} D_m^T u_m^T \otimes D_n^T B & 0 \\ -D_m^T u_m^T \otimes D_n^T B & I_{m-1} \otimes A \otimes I_{n-1} \end{bmatrix} \begin{bmatrix} g \\ \tilde{v}_0^T \end{bmatrix}$$  \hspace{1cm} (9)

If matrix $B$ satisfies $D_m^T B = 0$, it means that all the antennas are in the same plane. The ionospheric spatial gradient will then not be estimable. As a result, the DD observation equations will only contain the DD ambiguities as the unknown parameters, which is desirable for the estimation of the DD ambiguities since only the highly precise phase observables contribute to their estimation and precision. The number of observations, number of unknowns and the redundancy of the model of observations under $H_a$ are, respectively, $2f(m-1)(n-1)$, $f(m-1)(n-1)$ and $f(m-1)(n-1) + l$ and $f(m-1)(n-1) + l$, with $l$ being the dimension of the array (the number of estimable components of the ionospheric spatial gradient).
be computed as $P_{fa}$ based on the alarm rate parameter $l$. The procedure can also be followed, which means that $P_{fa}$ is normally distributed, i.e. $\hat{g}$ is obtained from the distribution $\mathcal{N}(g, Q_{\hat{g},\hat{g}})$. The variance matrix $Q_{\hat{g},\hat{g}}$ and, as a result, the MDB depend on several contributing factors like number of receivers, number of satellites, array configuration etc. Before conducting the actual measurements, one can decide on these contributing factors such that the requirements of the application at hand are satisfied. For instance, civil aviation precision approach needs to detect ionospheric spatial gradient greater than 300 mm/km with the probability of 0.9999 [14, 23, 26].

In the following, we will present analytical expressions for $Q_{\hat{g},\hat{g}}$ and the corresponding MDB. Numerical evidence will also be provided to support closed-form formulas.

### 3 Ionospheric spatial gradient UMPI MDB

#### 3.1 Single-epoch ambiguity-float scenario

To obtain closed-form expressions for $Q_{\hat{g},\hat{g}}$ and the corresponding MDB-vector $||g||d$, we start with the single-epoch ambiguity-float scenario. It can be shown that the variance matrix of the ionospheric spatial gradient estimator and its corresponding MDB, for 3-dimensional, 2-dimensional and 1-dimensional arrays reads as

$$||g|| = \sqrt{\frac{\lambda_0}{d^T Q_{\hat{g},\hat{g}} d}} \quad (11)$$

If the unit vector $d$ sweeps the surface of the unit sphere in $\mathbb{R}^l$, the whole MDB region is obtained. The shape and the orientation of the MDB region is governed by $Q_{\hat{g},\hat{g}}$, and its size is determined by $\lambda_0$. The MDB region is an interval when $l = 1$, an ellipse when $l = 2$, and an ellipsoid when $l = 3$.

The variance matrix $Q_{\hat{g},\hat{g}}$ and, as a result, the MDB depend on several contributing factors like number of receivers, number of satellites, array configuration etc. Before conducting the actual measurements, one can decide on these contributing factors such that the requirements of the application at hand are satisfied. For instance, civil aviation precision approach needs to detect ionospheric spatial gradient greater than 300 mm/km with the probability of 0.9999 [14, 23, 26].

In the following, we will present analytical expressions for $Q_{\hat{g},\hat{g}}$ and the corresponding MDB. Numerical evidence will also be provided to support closed-form formulas.

Figure 2. Chi-square non-centrality parameter $\lambda_0$ versus probability of missed detection $P_{md}$ for different values of probability of false alarm $P_{fa}$ when the degrees of freedom is (a) $l = 1$; (b) $l = 2$; (c) $l = 3$. 

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2.2 Testing procedure

To detect the ionospheric spatial gradient, we employ the UMPI test [20], as it results in the highest probability of the detection for a class of critical regions. The UMPI test statistic formed to detect and identify the ionospheric bias $g$ reads [20, 25]

$$T_l = g^T Q_{\hat{g},\hat{g}}^{-1} g \quad (10)$$

where $\hat{g}$ and $Q_{\hat{g},\hat{g}}$ are, respectively, the best linear unbiased estimator (BLUE) of $g$ and its corresponding variance matrix, computed under $H_0$ (that is Eq. (9)). The subscript $l$ shows the number of (functions of) estimable components of vector $g$. If the GNSS phase and code observables are normally distributed, $\hat{g}$ as a linear function of them is also normally distributed, i.e. $\hat{g} \sim \mathcal{N}(g, Q_{\hat{g},\hat{g}})$. The distribution of $T_l$ under $H_0$ is then the central Chi-square with $l$ degrees of freedom, i.e. $T_l \sim \chi^2_l(l, 0)$, and under $H_o$ the Chi-square with $l$ degrees of freedom and the non-centrality parameter $\lambda$, i.e. $T_l \sim \chi^2_l(l, \lambda)$, where $\lambda = g^T Q_{\hat{g},\hat{g}}^{-1} g$. Given the false alarm rate $P_{fa}$, if $T_l > \chi^2_{P_{fa}}(l, 0)$, then $H_o$ is rejected in favor of $H_o$. Here $\chi^2_{P_{fa}}(l, 0)$ is the critical value obtained from the distribution $\chi^2_l(l, 0)$ and chosen false alarm rate $P_{fa}$. The probability of missed detection can be determined by the use of $P_{fa}$, $\lambda$ and $l$ [20]. The reverse procedure can also be followed, which means that based on $P_{fa}$, $P_{md}$ and $l$, the non-centrality parameter can be computed as $\lambda_0 = \lambda(P_{fa}, P_{md}, l)$ visualized in Figure 2 [17, 20]. For example, when $P_{fa} = 10^{-4}$ and $P_{md} = 10^{-4}$, the non-centrality parameter would be $\lambda_0 = 57.9$, $\lambda_0 = 62.8$ and $\lambda_0 = 66.4$ for, respectively, $l = 1, l = 2$ and $l = 3$.

The boundary of the ellipsoidal MDB region is defined as $\lambda_0 = g^T Q_{\hat{g},\hat{g}}^{-1} g$ [20]. Parametrizing the MDB-vector as $g = ||g||d$, with unit $l$-vector $d$, the MDB can be computed as

$$||g|| = \sqrt{\frac{\lambda_0}{d^T Q_{\hat{g},\hat{g}} d}} \quad (11)$$

If the unit vector $d$ sweeps the surface of the unit sphere in $\mathbb{R}^l$, the whole MDB region is obtained. The shape and the orientation of the MDB region is governed by $Q_{\hat{g},\hat{g}}$, and its size is determined by $\lambda_0$. The MDB region is an interval when $l = 1$, an ellipse when $l = 2$, and an ellipsoid when $l = 3$.

The variance matrix $Q_{\hat{g},\hat{g}}$ and, as a result, the MDB depend on several contributing factors like number of receivers, number of satellites, array configuration etc. Before conducting the actual measurements, one can decide on these contributing factors such that the requirements of the application at hand are satisfied. For instance, civil aviation precision approach needs to detect ionospheric spatial gradient greater than 300 mm/km with the probability of 0.9999 [14, 23, 26].

In the following, we will present analytical expressions for $Q_{\hat{g},\hat{g}}$ and the corresponding MDB. Numerical evidence will also be provided to support closed-form formulas.
3-dimensional array

\[ Q_{ijk} = \frac{\sigma_p^2}{c_f^2 c_s^2} \times C_r^{-1} \]  \hspace{1cm} (12)

\[ ||g||_d = \frac{\sigma_p}{c_f c_s} \sqrt{\frac{\lambda_0}{d^T C_r d}}, \quad d \in \mathbb{R}^3 \]  \hspace{1cm} (13)

2-dimensional array

\[ Q_{ij} = \frac{\sigma_p^2}{c_f^2 c_s^2} \times \tilde{C}_r^{-1} \]  \hspace{1cm} (14)

\[ ||\tilde{g}||_d = \frac{\sigma_p}{c_f c_s} \sqrt{\frac{\lambda_0}{d^T \tilde{C}_r d}}, \quad d \in \mathbb{R}^2 \]  \hspace{1cm} (15)

1-dimensional array

\[ \sigma^2_g = \frac{\sigma_p^2}{c_f^2 c_s^2} \times \tilde{C}_r^{-1} \]  \hspace{1cm} (16)

\[ ||\tilde{g}||_d = \frac{\sigma_p}{c_f c_s} \sqrt{\frac{\lambda_0}{\tilde{C}_r}} \]  \hspace{1cm} (17)

where

\[ c_f = \sqrt{f(\mu^2 + \sigma_p^2)}, \quad c_s = \sqrt{w(1 - \frac{w^2}{m^2})} \]  \hspace{1cm} (18)

\[ C_r = (B - e_n b_n^T)(B - e_n b_n^T)^T \]  \hspace{1cm} (19)

\[ \tilde{C}_r = [u,v]^T C_r [u,v], \quad \tilde{C}_r = [u]^T C_r [u] \]  \hspace{1cm} (20)

with \( b_n^T = \frac{1}{3} e_n^T B, \mu = \frac{1}{7} \sum_{j=1}^f \mu_j, \sigma_p^2 = \frac{1}{7} \sum_{j=1}^f (\mu_j - \mu)^2, \) \( \bar{\mu} = \frac{1}{7} \sum_{j=1}^m w_j, \) and \( u, v \) forming an orthonormal basis of the array-plane such that \( u \) spans the linear array. \( ||\tilde{g}||_d \) is the norm of the projection of \( g \) onto the array-plane spanned by \( u, v \), and \( ||\tilde{g}||_d \) is the absolute value of the projection of \( g \) onto the linear-array spanned by \( u \). Be aware that the ionospheric spatial gradient component(s) orthogonal to the array is (are) not estimable.

Eq. (19) shows that both the variance matrix and the corresponding MDBs are invariant for our choice of the reference receiver, and hence how receiver pairs are selected to form the baselines. From Eqs. (12), (14) and (16), we do see that the precision of the ionospheric spatial gradient is dominated by code observables. That is because in single-epoch ambiguity-floating scenario, all the DD phase observables are preserved for the DD ambiguities. Therefore, the estimator of the ionospheric spatial gradient, the so-called float-solution, does not benefit from the high phase precision.

The main difference between the MDB regions of different types of arrays lies in \( \lambda_0 \) (because of different values for \( l \)) and the array configuration. Based on the analytical expressions derived above, we conduct a pre-analysis on the role played by various contributing factors in the MDB region of ionospheric spatial gradient. All the conclusions are valid for all types of arrays. We also present our conclusions visually for a planar array, which can also be illustrated for other types of arrays in a similar way. Only for simplicity, we consider a circle to put antennas over and make use of polar coordinates with respect to its center.
Code-precision-specific contribution: There is a direct relation between the MDB and $\sigma_p$, such that an increment in $\sigma_p$ will result in MDB region growing. Figure 3 depicts the impact of $\sigma_p$ on the MDB region, for the shown planar array and satellite geometry. The coordinates of antennas with respect to the center of circle have the form of $b_r = [r \cos \theta, r \sin \theta]^T$. Since the geometry of the array is symmetric, the MDB region would be of circle shape. The use is made of frequencies $L1$ and $L2$. $P_{fa}$ and $P_{md}$ should be set based on the requirements of the specific application at hand. For civil aviation applications, 'however, a standardized way of determining such a probability has not yet been developed and needs to be investigated' [23]. Here, we choose $P_{fa} = 10^{-4}$ and $P_{md} = 10^{-4}$, consistent with the previous works [13, 14, 27]. The graphs in this figure have been plotted for two extreme cases, namely the signals from the highest PRN10 and the lowest PRN28 satellites are contaminated by the ionospheric front. The linear dependency between the MDB size and $\sigma_p$ is visible in this figure.

$g_E$ and $g_N$ in Figure 3 (and all the figures coming after) are, respectively, the projection of $g$ on the East and North direction.

Non-centrality-specific contribution: The MDB is proportional to the square-root of $\lambda_0$. From Figure 2, for a given $P_{fa}$, $\lambda_0$ will get higher values for smaller $P_{md}$ and larger $p$. Larger $\lambda_0$ would result in larger MDBs.

Satellite-specific contribution: GNSS satellites effect has three aspects, which are number of satellites $m$, satellites elevation through the mean elevation-dependent weight $w^e$, and elevation of the contaminated satellite through the elevation-dependent weight $w^c$. Assuming that $w^1 \approx \ldots \approx w^m \approx w^e$, it can be seen that the number of satellites affects the MDB by approximately $\sqrt{m/(m-1)}$. Since the growth speed of $\sqrt{m/(m-1)}$ is much less than that of $m$, the rate of change of MDB as function of number of satellites will not be very considerable after adding a particular number of satellites. Figure 4 shows the behavior of MDB for different numbers of satellites, for the planar array and satellite geometry in Figure 3. The MDB region experiences a notable decrease when going from $m = 2$ to $m = 4$. As more other satellites are included, the improvement of MDB decreases.

Satellite elevations manifest themselves through the elevation-weighting $w^e$, which are reversely related to the satellite elevation. Two questions may be asked here: 1) what is the best constellation to detect ionospheric spatial gradient? 2) given a specific constellation, on which satellite, ionospheric spatial gradient can be detected the best? To answer the first question, we consider the following equivalent expression for $c_s$:

$$c_s = \left( \frac{1}{w^e} + \frac{1}{\sum_{c=1, c\neq e}^{m} w^c} \right)^{\frac{1}{2}}$$  \hspace{1cm} (21)

based on which, one can easily conclude that the maximum value for $c_s$ is obtained when all the visible satellites are in zenith, i.e. $e^s = 90^\circ$ ($s = 1, \ldots, m$). Therefore, the best satellite configuration in terms of ionospheric spatial gradient detectability is the one containing zenith-elevated satellites.

When it comes to the second question, $m w^e$ in $c_s$ formula-
Table 1 shows the values of satellites have high elevations and some low elevations. Most of the satellites have almost high elevations, (b) most of the satellites have low elevations, (c) some of the satellites have high elevations and some low elevations. The worst detectability is associated with those satellites among the satellites in view, that satellite of which the ionospheric front affects, the higher the elevations of the satellites in view, the better the behavior of $c_t$ with respect to changes in $w^e$. There-fore, the best detectability of the ionospheric spatial gradi-ent would be realized when the ionospheric front affects, among the satellites in view, that satellite of which the elevation-dependent weight is the closest to $\sum_{e=1,\neq c} w^e$. The worst detectability is associated with those satellites of which the elevation-dependent weights are the furthest from $\sum_{e=1,\neq c} w^e$. Figure 5 illustrates 3 GPS satellite configurations when (a) most of the satellites have almost high elevations, (b) most of the satellites have almost low elevations, (c) some of the satellites have high elevations and some low elevations. Table 1 shows the values of $\sum_{e=1,\neq c} w^e$ and $c_t$ for these configurations when the lowest and highest satellites are contami-nated. We can see that within each satellite configuration, $\sum_{e=1,\neq c} w^e$ gets its largest value when the highest satellite is corrupted by the ionospheric front. Moreover, the higher the elevations of the satellites in view, the better the ionospheric spatial gradient would be detectable.

Frequency-specific contribution: Three parameters represent the influence of the GNSS signal frequencies on the MDB, which are number of frequencies $f$, squared mean of the frequencies ratios $p^2$, and the dispersion of the frequencies ratios $\sigma_p^2$. The more frequencies are used, the MDB region will decrease more. If we go from single-frequency ($L_1$) to dual-frequency ($L_1 + L_2$) model of observations, then $p^2$ increases from 1 to almost 1.7, and $\sigma_p^2$ grows from 0 to almost 1. Therefore, the MDB region corresponding with the dual-frequency scenario is nearly 2 times smaller than that corresponding with the single-frequency scenario. It is shown in Figure 6.

Array-specific contribution: The contribution of the array is characterized through the number of receivers $n$ and the array configuration $B$. As the number of receivers increases, the MDB region gets smaller in all or some particular directions depending on the array geometry. Figure 7 illustrates the MDB regions as function of number of receivers for the satellite geometry in Figure 3.
A symmetric array geometry satisfies $D$ by the columns of differencing matrix $\theta$.

An equivalent form for $C$ the array configuration and satellite geometry in Figure 3.

If the length of the array increases, then the term $C_{\theta}^{-1}$ decreases, so does the MDB region. In Figure 8, the MDB region has been plotted for different lengths of array for increases, so does the MDB region. In Figure 8, the MDB region has been considered to be contaminated by the ionospheric front.

![Figure 6](image)

**Figure 6.** The ellipses showing the single-epoch ambiguity-float MDBs corresponding with the planar array and satellite geometry in Figure 3, with $m = 6, n = 4, \sigma_{fa} = 10^{-4}$ and $P_{md} = 10^{-4}$ for different number of frequencies. Solid and dashed curves are, respectively, associated with $\sigma_{f} = 0.2$ meter and $\sigma_{f} = 1$ meter. The lowest satellite PRN28 with the elevation $\varepsilon = 30.7^\circ$ has been considered to be contaminated by the ionospheric front.

If the length of the array increases, then the term $C_{\theta}^{-1}$ decreases, so does the MDB region. In Figure 8, the MDB region has been plotted for different lengths of array for the array configuration and satellite geometry in Figure 3. An equivalent form for $C_{\theta}^{-1}$ can be obtained as

$$C_{\theta}^{-1} = (B^T B)^{-1} + \frac{(B^T B)^{-1} B^T e_n e_n^T B (B^T B)^{-1}}{n - e_n^T P_{fa} e_n}$$

with $P_{fa} = B (B^T B)^{-1} B^T$. The smaller $e_n^T B$ (the projection of matrix $B$ onto the vector $e_n$), the smaller $C_{\theta}^{-1}$. If $e_n^T B = 0$, then the second term in Eq. (22) will disappear. A symmetric array geometry satisfies $e_n^T B = 0$, hence being desirable in terms of the MDB region size. The vector $e_n$ is the orthogonal complement of the space formed by the columns of differencing matrix $D_n$, i.e. $D_n^T e_n = 0$.

Therefore, the smaller $D_n^T B$ (the projection of matrix $B$
onto the range space of $D_n$, the larger $e_m^TB$, and the larger $C_{r}^{-1}$. The term $D_m^TB$ gets its small values when the baseline vectors $b_r$ ($r = 1,\ldots,n$) are close to each other. In Figure 9, the MDB regions for different geometries of 4 antennas, and the satellite configuration in Figure 3 are illustrated. In Figure 9 (b), the shape of antennas is almost aligned with the East direction, which is why the corresponding ionospheric spatial gradient MDB ellipse has its minor axis parallel to the East direction. In fact, this shape of array enables the user to find smaller gradients in East direction. The MDB area associated with Figure 9 (c) can be interpreted analogously.

From Figures 3 to 9, it can be seen that achieving small MDBs for the ionospheric spatial gradient with test in Eq. (10) with $P_{fa} = 10^{-4}$ and $P_{pred} = 10^{-4}$ based on the single-epoch code observables (ambiguity-float scenario) would require a large number of receivers or long distances between antennas. The above MDBs might be desirable for some special applications, but not for civil aviation precision approach. For such an application, due to the airport geometry and the high price of the receivers, increasing the number of receivers or the length of the array may not always be possible. So, does it mean it is impossible to detect ionospheric spatial gradient at the level of 300 mm/km upon the single-epoch observations? We answer this question in the following.

3.2 Single-epoch ambiguity-fixed scenario

Once the DD ambiguities are fixed, the phase observables would also contribute to the estimation of the ionospheric spatial gradient, hence improving its precision and reducing the MDB region size. The variance matrix of the ionospheric spatial gradient estimator after fixing the DD ambiguities, the so-called fixed solution, and the corresponding MDB can be shown to have the following form for the 3-dimensional array

$$Q_{\hat{g}_g} = \frac{\varepsilon}{1 + \varepsilon} Q_{\hat{g}_g}$$

(23)

$$||g||d = \sqrt{\frac{\varepsilon}{1 + \varepsilon} \sigma_p^2 \frac{\lambda_0}{d^T C_{r} d}}, \; d \in \mathbb{R}^3$$

(24)

where $\varepsilon = \frac{\sigma_0^2}{\sigma_p^2} \approx 10^{-4}$. For other types of array, the same relationship between float and fixed parameters holds true. Comparing Eqs. (23)/(24) and (12)/(13), if one can fix the DD ambiguities successfully at a single epoch, then the precision of ionospheric spatial gradient estimator improves by $10^4$, and the size of the corresponding MDB region reduces by $10^2$. Therefore, it is important to know how it would be possible to fix the DD ambiguities with an acceptable success rate at only one epoch. The success rate of the integer DD ambiguity resolution depends on their variance matrix. It can be shown that the variance matrix of the DD ambiguities for single-epoch scenario based on Eqs. (9) (fault-affected model $H_a$) and (2) reads

$$Q_{g_{\hat{g}}_{g_{\hat{g}}}} = \frac{\sigma_e^2}{C_g} \frac{D_m^T W^{-1} D_m \otimes \Lambda^{-1} \otimes D_m^T D_m}{C_p}$$

(25)

This variance matrix of order $f(m - 1)(n - 1)$ has two terms. The first term $C_g$ is dominated by high phase precision, while the second one $C_p$ is dominated by poor code precision. $C_p$ is due to the presence of the ionospheric spatial gradient in model $H_a$. In other words, based on the model of observations in Eq. (4) (fault-free model $H_o$), $C_p$ would disappear. In that case, the precision of the DD ambiguities would be at the level of the DD phase precision. Based on Eq. (25), we can analyze the efficacy of various factors on the precision of the DD ambiguities estimators and the consequent integer ambiguity resolution success rate. But, since the only purpose of fixing the ambiguities is to improve the precision of the ionospheric spatial gradient, it is important to know how setting these factors in favor of the ambiguity resolution, affects the variance matrix of the fixed ionospheric spatial gradient estimator $Q_{\hat{g}_g}$. Looking at Eqs. (23) and (25), $Q_{g_{\hat{g}}_{g_{\hat{g}}}}$ and $Q_{\hat{g}_g}$ are both dependent on $Q_{\hat{g}_g}$. The variance matrix of the fixed solution $Q_{\hat{g}_g}$ is a ratio of that of the float solution $Q_{\hat{g}_g}$, and
the illustrated array configurations, for different number of frequencies and different elevations of the contaminated satellite when \( \sigma_p = 0.2 \) meter. It can be seen when the GNSS observations are made on 2 frequencies \((L1 + L2)\), the DD ambiguities can be fixed by 100\% regardless of array and satellite configuration, if the elevation of the contaminated satellite is sufficiently high. When the elevation of the contaminated satellite is low, then both array and satellite configurations become important. If the GNSS observations are available on only 1 frequency \((L1)\), it is impossible to fix the ambiguities successfully at a single epoch.

Figure (10) (a) and (b) shows MDB regions corresponding to array configurations (2) and (1) in Table 2, respectively, for dual-frequency ambiguity-float scenario. MDB region associated with the array configuration (1) is indeed better than the one corresponding to the array configuration (2), since it is symmetric and its maximum boundary is 10 times smaller. Based on the results in Table 2, it is possible to fix the DD ambiguities associated with the array configuration (2) and satellite configuration (a) at one epoch. Figure (10) (c) illustrates the fixed MDB region. Although it is not symmetric, its maximum boundary is 10 times smaller than that of the float MDB region associated with the array configuration (1). This figure tells us, for some special satellite geometries, it is possible to detect the ionospheric spatial gradient even smaller than 300 mm/km at one single epoch when the shape of the array is to the benefit of the successful ambiguity resolution and the GNSS observables are available on 2 frequencies. For other satellite geometries that the high success rates are not achievable at one epoch, only a few number of epochs are needed to reach a desirable success rate.
Table 2. DD ambiguity resolution ILS success rate Ps (%) as function of the satellite configurations in Figure 5 and the array configurations shown, number of frequencies $f$, and the elevation of the contaminated satellite $\xi^c$ for $\sigma_p = 0.2$ meter.

<table>
<thead>
<tr>
<th>Array configuration $e^c$</th>
<th>Satellite configuration (a) $f = 1$</th>
<th>Satellite configuration (b) $f = 2$</th>
<th>Satellite configuration (c) $f = 1$</th>
<th>Satellite configuration (c) $f = 2$</th>
</tr>
</thead>
<tbody>
<tr>
<td>$r = 500$ meter $\theta = 0^\circ$, $90^\circ$, $180^\circ$, $270^\circ$</td>
<td>Highest $Ps$ (%) $10.921$</td>
<td>$100$</td>
<td>$9.962$</td>
<td>$99.999$</td>
</tr>
<tr>
<td>$r = 500$ meter $\theta = 0^\circ$, $10^\circ$, $20^\circ$, $70^\circ$</td>
<td>Highest $Ps$ (%) $65.660$</td>
<td>$100$</td>
<td>$61.135$</td>
<td>$100$</td>
</tr>
</tbody>
</table>

Table 3. DD ambiguity resolution ILS success rate Ps (%) as function of the satellite configurations in Figure 5 and the array configurations shown, and the elevation of the contaminated satellite $\xi^c$ for $f = 2 (L1 + L2)$ and $\sigma_p = 1$ meter.

<table>
<thead>
<tr>
<th>Array configuration $e^c$</th>
<th>Satellite configuration (a) $f = 1$</th>
<th>Satellite configuration (b) $f = 2$</th>
<th>Satellite configuration (c) $f = 1$</th>
<th>Satellite configuration (c) $f = 2$</th>
</tr>
</thead>
<tbody>
<tr>
<td>$r = 500$ meter $\theta = 0^\circ$, $90^\circ$, $180^\circ$, $270^\circ$</td>
<td>Highest $Ps$ (%) $20.014$</td>
<td>$16.903$</td>
<td>$18.836$</td>
<td></td>
</tr>
<tr>
<td>$r = 500$ meter $\theta = 0^\circ$, $10^\circ$, $20^\circ$, $70^\circ$</td>
<td>Highest $Ps$ (%) $50.578$</td>
<td>$44.726$</td>
<td>$48.171$</td>
<td></td>
</tr>
</tbody>
</table>

Table 3 contains the ILS success rate for the satellite configurations in Figure 5 and the illustrated array configurations, for $f = 2 (L1 + L2)$ and different elevations of the contaminated satellite when $\sigma_p = 1$ meter. It shows that for such a large value of $\sigma_p$, it would be impossible to reach high values for ILS success rate even if the GNSS observables are available on 2 frequencies.

Up to now, we have assessed the case of detecting iono-
spheric spatial gradient upon the single-epoch GNSS observables. From the results we obtained above, fixing the DD ambiguities for single-epoch scenario to achieve the desirable MDB regions depends on some factors like the satellite geometry and the elevation of the contaminated satellite which are uncontrollable and probably hinder having acceptable DD ambiguity resolution success rate. It is also known that, since the ambiguities are considered constant over time, if we let the time go on and collect GNSS observables over a particular number of epochs, more precise estimations for the DD ambiguities will be obtained, giving rise to larger success rates. In the following subsection we will address the multi-epoch scenario.

3.3 Multi-epoch scenario

When it comes to multi-epoch scenario, one can think of two cases: 1) ionospheric spatial gradient is time-invariant; 2) ionospheric spatial gradient is time-variant. In Table 4, the variance matrices of the ionospheric spatial gradient estimator corresponding with epoch $i$ are illustrated as function of $Q_{\hat{g}}$ in Eq. (12) specified for epoch $i$; for several scenarios and the two cases time-invariant and time-variant ionospheric spatial gradient. The closed-form formulations for the MDB can be easily found by substituting the variance matrices into Eq. (11). Although the variance matrix notation in this table is the one used for the 3-dimensional array, the results in this table holds true for all types of array.

3.3.1 Time-invariant ionospheric spatial gradient

First we consider the case when the ionospheric spatial gradient is time-invariant. When the ambiguities are still float, including the observations of more than one epoch, say $k$ epochs, improves the variance matrix of $\hat{g}$ by almost $k$ factor, and the MDB region by $\sqrt{k}$. Phase observables have no contribution to the variance matrix of $\hat{g}$ unless the integer DD ambiguities are successfully fixed. The reason for this can be explained as follows. We have assumed that both $g$ and $\sqrt{\lambda_{\hat{g}}}$ are constant over time. Therefore, the time-averaged DD phase observations are reserved for the integer-valued DD ambiguities, and only the time-differenced DD phase observations remain to contribute to the solution of the ionospheric spatial gradient. This contribution would, however, be zero since the time-differenced design matrix of the ionospheric spatial gradient is zero, and also phase observables are uncorrelated with the code ones. When the ambiguities are fixed, DD phase observables act as if they are highly precise DD code observables. Hence, both phase and code observables would contribute to the estimation and the corre-

### Table 4

<table>
<thead>
<tr>
<th>Scenario</th>
<th>Iono. gradient</th>
<th>Ambiguity</th>
<th>Variance matrix</th>
</tr>
</thead>
<tbody>
<tr>
<td>Single-epoch</td>
<td>-</td>
<td>float</td>
<td>$Q_{\hat{g}}(i)$</td>
</tr>
<tr>
<td>Single-epoch</td>
<td>-</td>
<td>fixed</td>
<td>$\frac{1}{1 + k} Q_{\hat{g}}(i)$</td>
</tr>
<tr>
<td>multi-epoch</td>
<td>time-invariant</td>
<td>float</td>
<td>$\sum_{b=1}^{m} [Q_{\hat{g}}(b)]^{-1}$</td>
</tr>
<tr>
<td>multi-epoch</td>
<td>time-invariant</td>
<td>fixed</td>
<td>$\frac{1}{1 + k} \left[ \sum_{b=1}^{m} [Q_{\hat{g}}(b)]^{-1} \right]$</td>
</tr>
<tr>
<td>multi-epoch</td>
<td>time-variant</td>
<td>float</td>
<td>$\frac{1}{1 + k} (1 + \frac{1}{k} Q_{\hat{g}}(i))$</td>
</tr>
<tr>
<td>multi-epoch</td>
<td>time-variant</td>
<td>fixed</td>
<td>$\frac{1}{1 + k} Q_{\hat{g}}(i)$</td>
</tr>
</tbody>
</table>
The precision of the DD ambiguities would benefit from the number of epochs. It can be shown that the variance matrix of the DD ambiguities after \( k \) epochs has the following form

\[
Q_{\hat{g}\hat{g}} = \sigma_p^2 \sum_{i=1}^{4} (D_{w}^{g}w^{-1}(i)D_{w}^{g})^{-1} \otimes \Lambda^{-1} \otimes D_{w}^{g}D_{w}^{g} + D_{w}^{g}w^{-1}D_{w}^{g} \otimes \Lambda^{-1} \otimes D_{w}^{g}B^{2} \sum_{i=1}^{4} (Q_{\hat{g}\hat{g}}^{-1}(i))^{-1} B^{2} D_{w}^{g}.
\]

This shows that \( Q_{\hat{g}\hat{g}} \) will improve by almost \( k \) factor after \( k \) epochs. This means that if the values of some uncontrollable factors such as satellites configuration and the number of satellites does not provide us with the required single-epoch precision for the DD ambiguities, then letting the time go on would be of help in that sense.

From Table 3, we saw that when \( \sigma_p = 1 \) meter, it is impossible to reach high ILS success rates for the ambiguity resolution based on single-epoch observables. In Table 5, one can see the number of epochs (with the sampling-rate of 30 seconds) needed for \( P_s = 99.99999\% \) when the ionospheric spatial gradient is time-invariant, for the satellite configurations in Figure 5 and the illustrated array configurations, for \( f = 2 \) \((L1 + L2)\) and different elevations of the contaminated satellite when \( \sigma_p = 1 \) meter.

### 3.3.2 Time-variant ionospheric spatial gradient

Now we consider the case where the ionospheric spatial gradient is assumed to change over time. In order to derive closed-form expressions for this case, we have made the assumption that over a limited number of epochs, say \( k \), \( w^i(1) \approx \ldots \approx w^i(k) \approx w^i \). The variance matrix of \( \hat{g}(i) \) will not be improved by including other epochs observables, in case only code observables are taken into account. That is because, according to Eq. (9), the ionospheric spatial gradient vectors are estimated based on the time-uncorrelated DD code observables and observation models with no common parameters. When the phase observables are also included and the ambiguities are float, the variance matrix of the ionospheric spatial gradient and the corresponding MDB drops by \( k \) and \( \sqrt{k} \), respectively. When the ambiguities are successfully fixed, DD phase observables act as highly precise DD code observables. In this case, the ionospheric spatial gradient vectors are estimated based on the time-uncorrelated code and phase observables and observation models with no common parameters. So, there would be no correlation among the estimators of the ionospheric spatial gradient associated with various epochs.

Assuming that \( w^i(1) \approx \ldots \approx w^i(k) \approx w^i \), the variance matrix of the DD ambiguities after \( k \) epochs can be approxi-

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**Table 5.** Number of epochs needed for the DD ambiguity resolution ILS success rate to reach 99.9999%, as function of the satellite configurations and the array configurations shown, for \( f = 2 \) \((L1 + L2)\) and \( \sigma_p = 1 \) meter. The data sampling-rate is 30 seconds.

<table>
<thead>
<tr>
<th>Array configuration</th>
<th>( r = 500 ) meter</th>
<th>( \theta = 0^\circ, 90^\circ, 180^\circ, 270^\circ )</th>
<th>( \theta = 0^\circ, 10^\circ, 20^\circ, 70^\circ )</th>
</tr>
</thead>
<tbody>
<tr>
<td>Highest time-invariant</td>
<td>24</td>
<td>28</td>
<td>24</td>
</tr>
<tr>
<td>time-variant</td>
<td>25</td>
<td>29</td>
<td>25</td>
</tr>
<tr>
<td>Lowest time-invariant</td>
<td>44</td>
<td>55</td>
<td>44</td>
</tr>
<tr>
<td>time-variant</td>
<td>43</td>
<td>54</td>
<td>45</td>
</tr>
</tbody>
</table>

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mated by
\[
Q_{\hat{z}}^{k} \approx \frac{\sigma^2}{k} \delta^T W^{-1} D_n \otimes \Lambda^{-2} \otimes D_n^T D_n \\
+ \frac{1}{k} \delta^T \mu \mu^T \Lambda^{-1} \otimes D_n^T B Q_{\tilde{z}}^{-1} B^T D_n
\]

which clearly shows \(Q_{\hat{z}}^{k} \) will decrease by almost \(k\) after \(k\) epochs.

Table 5 shows the number of epochs needed for the DD ambiguity resolution ILS success rate to reach \(P_s = 99.9999\%\) when the ionospheric spatial gradient is time-invariant, for \(f = 2\) (\(L_1 + L_2\)) and \(\sigma_p = 1\) meter, as function of the array configurations, the satellite configurations, and the elevation of the contaminated satellite.

According to Table 5, fixing the DD ambiguities does not have a dependency on whether the ionospheric spatial gradient is time-invariant or time-variant. This conclusion has already been confirmed by comparing Eqs. (27) and (28). The reason for this can be explained as follows. In both cases, ambiguities at epoch \(k\) are estimated based on the time-averaged phase observables, corrected for the ionospheric spatial gradient estimator at epoch \(k\). Looking at Table 4, one can see that the variance matrices associated with epoch \(k\) of the time-invariant ionospheric spatial gradient and the time-variant one are approximately the same. As long as the ambiguities are flat, time-variant ionospheric spatial gradient is estimated using code observables (uncorrelated with the phase measurements), and time-invariant ionospheric spatial gradient is estimated using code and time-differenced phase observables (uncorrelated with the time-averaged phase measurements). Therefore, the precisions of the ambiguities are almost the same for both cases.

4 Summary and conclusions

The detectability of the ionospheric spatial gradient using the test in Eq. (10) depends on various factors summarized below in bullets.

- Increasing the number of receivers and the length of the array would lead to detecting smaller ionospheric spatial gradients with the test in Eq. (10). Regarding the array geometry, we showed that configurations fulfilling \(e_f^T B = 0\) will decrease the size of the UMPI MDB. However, if one aims at fixing the DD ambiguities with a high success rate at only one epoch, this array geometry would not be of much help. High ILS success rates will be obtained if the array geometry satisfies \(D_f^T B \rightarrow 0\), which is not desirable for the MDB size. At the same time, one should keep in mind that if the DD ambiguities are successfully fixed, then the highly precise phase observables will contribute to the precision of the ionospheric spatial gradient, making the MDB size very small.

- Both the variance matrix and the MDB region of the ionospheric spatial gradient are independent of our choice of the reference receiver and reference satellite.

- Increasing the number of satellites will give rise to enhancing the ability to detect smaller ionospheric spatial gradients. The higher the elevation of the satellites in view, the better the detectability of the ionospheric spatial gradient will be. Given a satellite configuration, the best ionospheric spatial gradient detectability will be obtained when the ionospheric front affects satellite \(c\) of which the elevation-dependent weight \(w^c\) is the closest to \(\sum_{i=1,i\neq c} w^i\). The behavior of the precision of the DD ambiguities with respect to the number of satellites and the satellites elevation can be explained the same way.

- Having GNSS observables on more than one frequency available would be of much help in terms of reducing the minimal detectable ionospheric spatial gradient.

- Involving the observables of more than one epoch, say \(k\) epochs, has different effects for different scenarios. When the ionospheric spatial gradient is considered to be time-invariant, as long as the ambiguities are float, only the code observable of \(k\) epochs contribute to the estimation of the ionospheric spatial gradient, improving its precision by almost \(k\) factor and reducing the corresponding MDB size by \(\sqrt{k}\). Once the DD ambiguities are successfully fixed, phase observables also play role in the precision of the ionospheric spatial gradient, and enhance it by \(\varepsilon = 10^{-4}\) and the associated MDB by \(\sqrt{\varepsilon}\) compared with the ambiguity-float scenario.

- When the ionospheric spatial gradient is assumed to change over time, its precision at each epoch will not benefit from observables of other epochs, if only the code observables are included. As the phase measurements are taken into account, and ambiguities are float, the precision and the MDB improve by \(k\) and \(\sqrt{k}\), respectively. Once the ambiguities are fixed, the precision and the MDB region enhance by, respectively, \(\varepsilon\) and \(\sqrt{\varepsilon}\) with respect to multi-epoch code-only scenario.
Based on the numerical results obtained, when the ambiguities are still float, even if there are 8 symmetrically-distributed receivers with the length of 500 meter, and GNSS observables are available on 2 (L1 and L2) frequencies, the ionospheric spatial gradient smaller than 6000 mm/km and 1260 mm/km can not be detected with $P_{fa} = 10^{-4}$ and $P_{md} = 10^{-4}$ at single epoch when, respectively, $\sigma_p = 1$ meter and $\sigma_p = 0.2$ meter. The MDB region can become desirable by making the array larger. For example, if the length of the array (the radius of the array circle) reaches 2000 meter, then the previous numbers would reduce to 1500 mm/km and 315 mm/km.

Our outcomes showed that for an array configuration with 4 antennas satisfying $D_n^4 B \rightarrow 0$, the DD ambiguities could be fixed during a few number of epochs for dual-frequency scenario and $\sigma_p = 0.2$ meter. When $\sigma_p = 1$ meter, however, a large number of epochs are needed for the ILS success rate to reach a desirable value. For example, for a satellite geometry with highly-elevated satellites, when the data sampling-rate is 30 seconds, it would be 25 epochs, and for a satellite geometry with low-elevated satellites, it would be 37 epochs, to get ILS success rate more than 99.9999% when the ionospheric spatial gradient is assumed to change over time. If the ambiguities, by choosing an appropriate array configuration, could be fixed during a few number of epochs, the MDB size would reach values at the level of a few hundred mm/km.

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**References**


